SYSTEMS COMMAND



October 26, 2022 0830-1600 PDT

Controlled by: USSF Controlled by: SSC/CG CUI Category: N/A Distribution: Approved for Public Release: Distribution Unlimited POC: SSC/CG United States Space Force Space Systems Command MILCOMM & PNT Directorate

> Major Cobb Brandon Mr. Dan Godwin Captain Andrew Sweeten





LIVE MEETING LINK: https://saicwebconferencing.zoomgov.com/j/1603353663

Meeting ID: 160 335 3663 Passcode: 621616

DIAL IN: Dial by your location 571 200 1700 US Meeting ID: 160 335 3663 Passcode: 621616

In the event of Zoom problems during the discussion, please try and reconnect via zoom. If zoom continues to have issues beyond 15 minutes, please connect via MS Teams:

Alternate Meeting Application (only if needed)

Join on your computer or mobile app <u>https://dod.teams.microsoft.us/l/meetup-</u> join/19%3adod%3ameeting_c4dd5c38d0ae429191db25a307db6d5e%40thread.v2/0?context=%7b%22Tid%22%3a%228331b18d-2d87-48ef-a35fac8818ebf9b4%22%2c%22Oid%22%3a%2239eaebff-b71b-4aad-8a01-55fa5d59953e%22%7d

Toll number: +1 410-874-6740 Conference ID: 326 120 515# **Dial-in Information**



Roll Call

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Rules of Engagement





ABSOLUTELY NO PROPRIETARY, FOUO, CLASSIFIED, OR COMPETITION SENSITIVE INFORMATION IS TO BE DISCUSSED DURING THIS MEETING.



Rules of Engagement (Cont'd)

- Please place your phones on mute when not speaking to minimize background noise
- For dial-in attendees, DO NOT take calls from phone while on telecom
- Comments against the topics listed on the official agenda will get priority during discussion
- Topics that warrant additional discussion may be side-barred
- Walk-on topics may be discussed during the open discussion
- Meeting minutes and final Proposed Changes Notices (PCNs) will be generated and distributed as a product of this meeting
- Please announce your name and organization before addressing the group

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Rules of Engagement (Cont'd)

- Types of comments to be discussed/dispositioned:
 - Critical (C)
 - Substantive (S)
 - Rejected/Deferred Administrative (A)
- Comments are grouped by sub-topic rather than by comment type



Rules of Engagement (Cont'd)

The purpose of the meeting is to:

- 1) Obtain ICWG approval on the proposed language generated for the enterprise RFCs that impact the public documents
- 2) Discuss any new open forum items against the Public Signals in Space documents

SPACE

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Agenda

Public ICWG	Drecenter	
(1 st Half of Day)	Presenter	
GPS Public ICWG and Public Forum Meeting Overview and Roll Call, Rules of Engagement	Capt. Andrew Sweeten	8:30 8:45
Opening Remarks / GPS Overview	Maj. Stewart Brandon	8:45 9:05
2022 Public ICWG RFC Discussion		
 RFC-495 (2022 Proposed Changes to the Public Documents) 	Tony Anthony	9:05 11:05
Break		11:05 11:10
 Open RFC Discussion Session / Action Items 		11:10 11:30
Special Topics Presentations		
 Expected Almanac Behavior in OCX Era 	Dr. Andrew Hansen	11:30 12:00
Action Item Review		12:00 12:10

Public Forum (2 nd Half of Day)	Presenter	
Special Topics Presentations - Continu	ed	
Mitigate LNAV Default Nav Anomaly	Hamza Abusalam	13:00 13:20
CNAV Message Schedules	Bert Hayden	13:20 13:40
ISC and CEI Table	Karl Kovach	13:40 14:00
Data ID Issue	Karl Kovach	14:00 14:20
Break		14:20 14:35
Walk-on Topics, Open Discussion		14:35 15:00
PRAT - Public Req. Accountability Tool	Tony Anthony	15:00 15:30
Action Item Review		15:30 16:00
Closing Remarks	Maj. Stewart Brandon	16:00 16:10
Hot Wash (internal)		16:10 16:40

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Opening Remarks

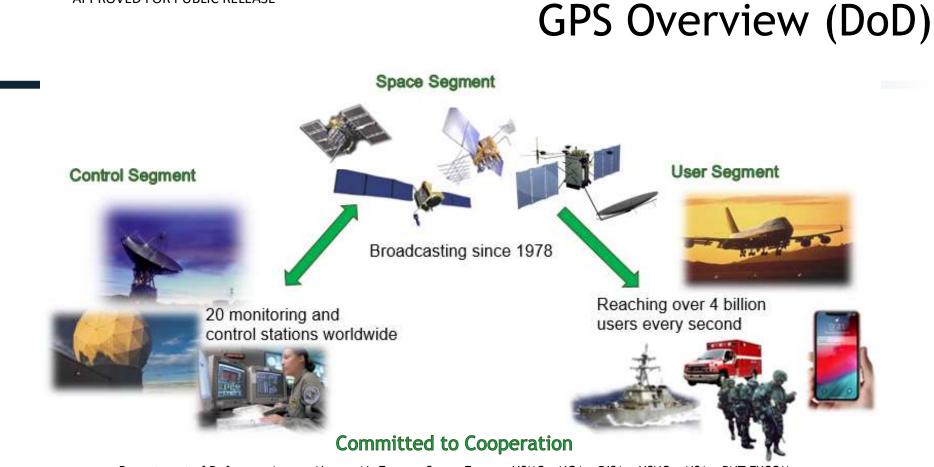
Position, Navigation, and Timing (PNT) Mission Area

October 26, 2022

Major Stewart C. Brandon Chief, Positioning, Navigation and Timing Requirements and Integration Branch

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Department of Defense • Army • Navy • Air Force • Space Force • USMC • NGA • DISA • USNO • NSA • PNT EXCOM National Nuclear Security Administration (NNSA) • Department of Transportation • Federal Aviation Administration Department of Homeland Security • U.S. Coast Guard • International Civil Aviation Organization Global Navigation Satellite Systems • Galileo • Beidou • GLONASS • QZSS • NAVIC International Committee on GNSS • International Telecommunication Union



Global Positioning Satellites: Encompassing the DoD and Civil Industry Partners

- GPS is utilized across the world with
- 6B+ users! GPS impacts almost every industry some of these industries include:
 - Agriculture
 - Maritime
 - Public Safety
 - Recreation
 - Space
 - Aviation
 - Finance
 - Telecommunications
 - Telematics
 - Oil/Gas
- GPS economic benefit ~\$1.4 Trillion*

GPS consistently met all technical performance commitments: Accuracy, Integrity, Availability and Continuity Integrity, Availability and Continuity

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GPS Overview



GPS Constellation Status

Baseline Constellation: 24 Satellites

Satellite Block	Quantity	Average Age (yrs)	Oldest
GPS IIR	12 (5*)	20.7	25.1
GPS IIR-M	8 (1*)	14.9	16.9
GPS IIF	12	8.6	12.3
GPS III	5	2.4	3.7

37 Satellites • 31 Set Healthy

*Not set healthy

As of 27 Aug 22

GPS Signal in Space (SIS) Performance

Week ending on 3 Sept 22

Average URE*	Best Day URE	Worst Day URE
49.1 cm	31.5 cm (20 Apr 21)	64.8 cm (20 May 22)

*All User Range Errors (UREs) are Root Mean Square values

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GPS Enterprise Roadmap

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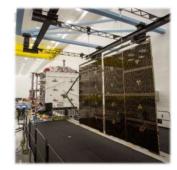


GPS III

- SV01 Set healthy and available for use on 13 Jan 20
- SV02 Set healthy and available for use on 1 Apr 20
- SV03 Set healthy and available for use on 1 Oct 20
- SV04 Set healthy and available for use on 2 Dec 20
- SV05 Set healthy and available for use on 25 May 22
- SV06 Launch scheduled for 18 Jan 23
- SV07 in storage AFL 20 May 21; TLD May 2024
- SV08 in storage AFL 10 Jun 21; TLD FY25
- SV09 in storage AFL 23 Aug 22; TLD FY26
- SV10 in production TLD FY26

Five GPS III satellites declared operational

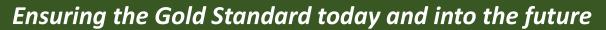






GPS III Follow-On (GPS IIIF)

- GPS IIIF additional features
 - Regional Military Protection (RMP) and redesigned Nuclear Detonation Detection System (NDS)
 - Search-and-Rescue (SAR) payload faster detection and location of distress signals
 - Laser Retroreflector Array (LRA) provides more precise ranging data
 - Partnering with Air Force Research Laboratory (AFRL) for future technology opportunities
 - Demo on Navigation Technology Satellite (NTS-3)
 - Digital Reprogrammable Payloads
 - Advanced Clocks
 - Status: Milestone C Completed 13 Jul 20; SV11 launch forecasted for FY2027



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Next Generation Operational Control System (OCX)

- Next-generation command, control and cyber-defense for GPS
 - Enhanced command and control capability
 - Modernized architecture
 - Robust information assurance and cyber security
- Incremental Development
 - OCX Block 0: Launch and Checkout System (LCS) for GPS III
 - OCX Blocks 1 and 2: Controls and manages all GPS IIR, GPS IIR-M, GPS IIF, and GPS III spacecraft; and controls all legacy and new GPS signals
 - OCX 3F: Adds support to OCX for GPS IIIF vehicle and new capabilities including Regional Military
 Protection
- Current Status
 - LCS successfully supported Launch and Checkout for GPS III SV01-SV05
 - OCX Block 1 completed factory integration and in Golden Dry Run for factory qualification
 - Constellation Transfer (CTX) 3QFY23; Operational Acceptance target 1QFY24

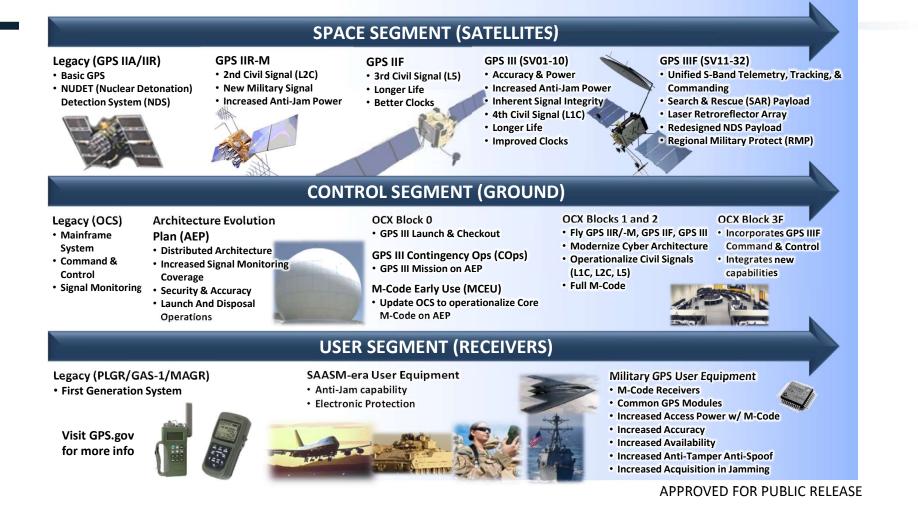
OCX program continues to execute and is nearing completion

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GPS Modernization





GPS Requirements Management

October 26, 2022

Major Stewart Brandon

Chief, Positioning, Navigation, and Timing Requirements and Integration Branch



GPS Requirements Team

Space Force

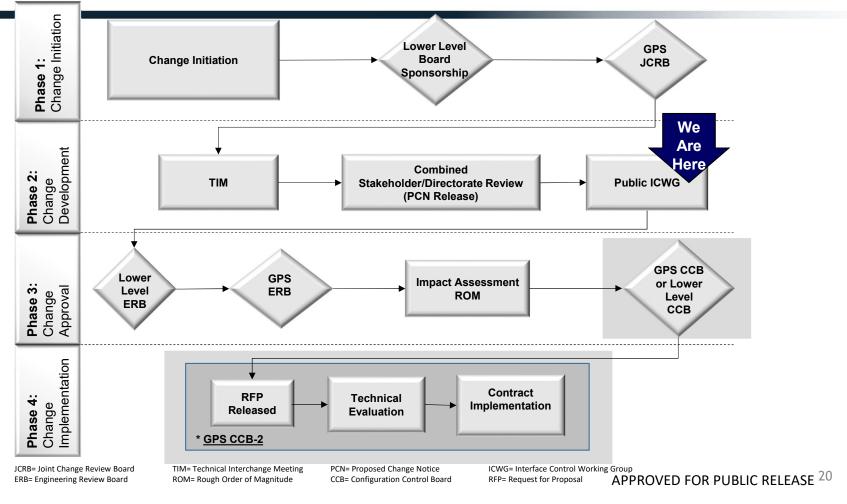
- Maj Stewart Brandon, Chief, PNT Requirements & Integration Branch
- Mr. Daniel Godwin, Chief, GPS Requirements Section
- Capt Adam Sweeten, GPS Requirements Action Officer

Aerospace

- Dr. Rhonda Slattery, Enterprise Requirements Lead
- Mr. Karl Kovach, Civil Requirements Lead
- Mr. Bert Hayden, Senior Engineer Specialist
- Systems Engineering and Integration (SE&I)
 - Mr. Don Latterman, Senior Technical Advisor
 - Mr. Tony Anthony, Responsible Engineer

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Technical Baseline Change Management Process Flow Chart





Action Items and Feedback

- We will record actions during the discussions and share during the Action Item agenda item
- If you have further actions or feedback after the 2022 PICWG please submit to smcgper@us.af.mil



QUESTIONS?



Positioning, Navigation, and Timing Mission Adjudication Working Group (AWG) Slides

PUBLIC-ICWG/AWG #2 26-OCT-2022

DOCUMENT CLASSIFICATION

Unclassified

REQUEST FOR CHANGE (RFC) NUMBER

RFC-495

RFC TITLE

2022 Proposed Changes to the Public Documents

GOVERNMENT POC Capt. Andrew Sweeten SSC/CGES, 310.653.4549

SE&I POC Tony Anthony, SSC/ZACS-PNT/SE&I, 310.418.7693X

CM POC

Veronica Quebedeaux, SSC/CGE/SE&I, 310.414.2856 APPROVED FOR PUBLIC RELEASE²³



RFC-495: 2022 Proposed Changes to the Public Documents

RFC CHANGE TYPE: Correction or Clarification to Baseline

1) PROBLEM STATEMENT:

- 1. Change Pconst to Rconst and MFDconst in the CNAV and CNAV2 Integrity Support Message. The rate of unalerted constellation failures (Rconst) and the mean duration of these failures (MFDconst) characterize such failures better than the probability of an unalerted constellation failure at any given time. (Pre-RFC-1200)
- 2. Implement Administrative Fixes needed on any document otherwise affected by any of the solutions to the above 5 problems.

2) SOLUTION:

- 1. Rework Pconst to Rconst and MFDconst in all affected documents
- 2. Provide clarity and clean up identified administrative changes in all affected documents IS-GPS-200, IS-GPS-705 and IS-GPS-800



RFC-495: 2022 Proposed Changes to the Public Documents

3) SF	PONSOR, D	RIVER		THIS RFC IS: RO	UTINE							
Sponsor: ZACS-PNT					Lead Driver Event	:: ICAO State Letter Revi	ew	Lead Driver Event Da	ate: June 2023 Dec 2	2022		
JC	RB T	ΪM	Stakeholder Review	Comments Due	Resolve Comments	AWG	GPS ERB	Impact Assessment Period	LL CCB	ССВ		
24 Ja	24 Jan 22 2 Feb 22 16 Feb 22 17 Feb 22		9 Jun 22 15 Mar 22 (CSR) 15 Jun 22 15 Jul 22 (Public Release)	25 Jul 22 14 Apr 22 30 Jul 22 19 Sep 22	30 Apr 22 15 Aug 22 5 Oct 22	11 May 22 (govt AWG) 28 Sep 22 26 Oct 22 (Public ICWG)	4 Jan 23 7 Dec 22	TBD	12 Jan 23 14 Dec 22	16 Jan 23 19 Dec 22		
4) A	PPROVAL A	UTHO	RITY: (Select or	ne)								
	Enterprise	Prop	Proposed Change (or Variance) affects multiple Segments' Technical Baseline(s), prime contractor documentation or external agencies outside of the GPS Directorate									
	Lower Level	Prop	roposed Change affects one Segments' Technical Baseline or prime contractor documentation									



Stakeholder Review Status - Combined Stakeholder Review

15) REVIEW STATUS

,	1	1	,		1						
Office	Response Required	No Impact	No Comments	Comments	No Response	EXTERNALSTAKEHOLDERS	Response Required	No Impact	No Comments	Comments	No Response
PCA (GPA)	RR		X			USSF, SpOC 2 SOPS (SPOC 2 SOPS)	RR		X		
ZACS-Civil (GPC)	RR		Х			Space Delta 8 (was 50OG)	RR		Х		
ZACS-NGA (GPD)	RR		X			PNT-MAT (was SpOC/S5M)	RR		X		
ZACS-PNT (GPE)	RR		X			National Security Agency	RR		Х		
PCET (GPEV)	RR				х	NGA	RR		X		
PCCC OCX 3F	RR			Х		PNT—PO (ADAP, MAGR, DAGR); EGI-M	RR		Х		
PCCC SMPS (GPGC)	RR	х				AFLCMC/EBD	RR				Х
PCCX (GPGX)	RR			х		SNL		х			
PCCN (GPGN)	RR		X			PRIME CONTRACTS					
PCL (GPL)	RR	х				IIR/IIR-M/III On Orbit Sust - LM	RR			х	
ECPG (GPL)	RR			х		IIA/IIF On-Orbit Sust - Boeing	RR		Х		
PCV (GPN; PCN)	RR			х		GPS III - LM	RR			х	
PCT (GPT)	RR				х	GPS IIIF - LM	RR			х	
PCU (GPU)	RR			х		GCS II – Sust - LM	RR	х			
PCM3 (GPV3)	RR			х		OCX 1/2 - RTX	RR				х
PCMF (GPV4)	RR			Х		OCX 3F - RTX	RR				Х
AEROSPACE	RR			х		MGUE Inc 1 - KTR A	RR		Х		
MITRE	RR			х		MGUE Inc 1 - KTR B	RR			х	
SE&I	RR	х				MGUE Inc 1 - KTR C	RR			х	
EXTERNAL STAKEHOLDER	s					MGUE Inc 2 MSI - BAE	RR			х	
PM-PNT (Army)	RR		See PCA (GP	A) Response		MGUE Inc 2 MSI - L3H	RR		X		
NIWC PAC	RR		See PCV (GP	N) Response		MGUE Inc 2 MSI - RTX	RR			х	
USNO	RR		See PCV (GPI	N) Response		SMPS Development - BAH	RR	х			
USMC	RR		See PCV (GP	N) Response		SMPS Sust - Lockheed Martin	RR		Х		
NAWCAD	RR		See PCV (GP	N) Response		USNDS Ground 6 / ICADS 7 - Sandia	RR	Х			



Public Review Comment Resolution Matrix (CRM) Status

CRM – COMBINED STA	CRM – COMBINED STAKEHOLDER/DIRECTORATE REVIEW STATUS:												
Disposition/Type	Critical	Substantial	Administrative	Totals	Concurrence	Notes							
Accept		35	1	36									
Accept with Comment		3	1	4									
Reject	2	3		5									
Deferred		1		1									
Grand Totals:	2	42	2	46									



Industry Comment Decisions

CRM – COMBINED STA	AKEHOLDER,	DIRECTORAT	E REVIEW STATU	S:		
Disposition/Type	Critical	Substantial	Administrative	Totals	Concurrence	Notes
Accept		9		9		
Accept with Comment		3	1	4		Industry Comments Europa
Reject	2			2		Thales
Deferred						• Aerospace Corp
Grand Totals:	2	12	1	15		



Change Constellation Terminology

COMMENT								
DOORS ID	IS200-2127, IS705-1741							
Paragraph	IS-GPS-200 30.3.3.10.1.8 IS-GPS-705 20.3.3.10.1.8	Comment Number	CRM #40, 41					
	Substantive	Disposition	Accept					
Comment Originator(s)	M. Mabilleau (Europa)							
Comment	The term "GNSS constellation" may be confusing as the ISM contain the parameter "GNSS ID". It is suggested to change "GNSS constellation" by "GNSS ID" or "constellation as per GNSS ID".							
Government Response	Agree							



Change Constellation Terminology

SYSTEMS COMMAND	
Paragraph	IS200-2127, IS705-1741
WAS	<inserted object=""></inserted>
Post AWG Proposed IS	Bits 78 through 81 of Message Type 40 shall provide the assumed mean duration of a satellite fault (MFD _{sat}) value for ARAIM at the current time for the associated GNSS constellation.
Newest Proposed IS	Bits 78 through 81 of Message Type 40 shall provide the assumed mean duration of a satellite fault (MFD _{sat}) value for ARAIM at the current time for the associated GNSS constellation as per GNSS ID.



Above The Allowed US Maximum Commitment

STEMS COMME									
DOORS ID	IS200-1789, IS705-1633								
Paragraph	IS-GPS-200 30.3.3.10.1.9 IS-GPS-705 20.3.3.10.1.9	Comment Number	CRM #42, 43						
	Substantive	Disposition	Reject						
Comment Originator(s)	M. Mabilleau (Europa)								
Comment		The range of R_{const} goes above the maximum commitment provided by US in GPS SARPs. Combined with some values from MFD _{const} , it could lead to P_{const} above the US maximum commitment for P_{const} in GPS SARPs.							
Government Response	This is a generic bit definition applicable to mo The commenter did not provide a specific sugg		nsider						



Constellation Fault Probability

OTSTEMS COMMAND	
Paragraph	IS200-1789
Paragraph of	The four bits are defined as follows:
Interest	$0000 = 3.16 \ge 10^{-34}$ /hour
	$0001 = 1 \ge 10^{-34}$ <u>/hour</u>
	$0010 = 3.16 \text{ x } 10^{-45} / \text{hour}$
	$0011 = 1 \ge 10^{-45}$ /hour
	$0100 = 3.16 \times 10^{-56}$ /hour
	$0101 = 1 \ge 10^{-56}$ /hour
	$0110 = 3.16 \text{ x } 10^{-67} / \text{hour}$
	$0111 = 1 \ge 10^{-67}$ /hour
	$1000 = 3.16 \text{ x } 10^{-78} / \text{hour}$
	$1001 = 1 \ge 10^{-78}$ /hour
	$1010 = 3.16 \times 10^{-89}$ /hour
	$1011 = 1 \ge 10^{-89}$ /hour
	$1100 = 3.16 \text{ x } 10^{-910} / \text{hour}$
	$1101 = 1 \ge 10^{-910}$ /hour
	$1110 = 3.16 \times 10^{-1011} / hour$
	1111 = RESERVED



Change Constellation Terminology

STATEMS COMMAN				
DOORS ID	IS200-1800, IS705-1647			
Paragraph	IS-GPS-200 30.3.3.10.1.10 (Corrected) IS-GPS-705 20.3.3.10.1.10	Comment Number	CRM #44, 46	
	Substantive	Disposition	Accept	
Comment Originator(s)	E. Canestri (Europa)			
Comment	The introduction of the notion of MFD _{const} instead of MFD, should be accompanied with a definition of what is meant by constellation fault.			
	"Value for ARAIM" is not deemed sufficient to understand what it meant with this parameter.			
	Add a note to clarify the MFDconst refers to the mean time between a fault exceeding the SPS SIS URE NTE tolerance from two or more satellites due to a common cause until the timely notification is issued to the user.			
Government Response	Agreed. SMEs provided language responding to	this comment.		



Define MFD_{const}

SYSTEMS COMMAND	
Paragraph	IS200-1800
WAS	Bits 82 through 85 of Message Type 40 shall provide the assumed Mean Fault Duration (MFD) value for ARAIM at the current time for the associated GNSS constellation.
Post AWG Proposed Redlines	Bits <u>8286</u> through <u>8589</u> of Message Type 40 shall provide the assumed <u>Meanmean</u> <u>Faultduration</u> <u>Duration of a constellation fault</u> (<u>MFDMFDconst</u>) value for ARAIM at the current time for the associated GNSS constellation.
Newest Proposed Redlines	Bits 8286 through 8589 of Message Type 40 shall provide the assumed Meanmean FaultdurationDuration of a constellation fault (MFDMFDconst) value for ARAIM at the current time for the associatedGNSS constellation.MFD _{const} is the mean time the instantaneous URE of two or more satellites exceed, due to a commoncause, 4.42 times the IAURA (or 5.73 times the IAURA when the conveying signal is provided with anenhanced level of integrity assurance) without a timely notification issued to the user.



Change Constellation Terminology

DOORS ID	IS200-1788, IS705-1632			
Paragraph	IS-GPS-200 30.3.3.10.1.9 IS-GPS-705 20.3.3.10.1.9	Comment Number	CRM #45, 47	
	Substantive	Disposition	Accept	
Comment Originator(s)	E. Canestri (Europa)			
Comment	The introduction of the notion of MFD _{const} instead of MFD, should be accompanied with a definition of what is meant by constellation fault.			
	"Value for ARAIM" is not deemed sufficient to understand what it meant with this parameter.			
	Add a note to clarify the MFDconst refers to the mean time between a fault exceeding the SPS SIS URE NTE tolerance from two or more satellites due to a common cause until the timely notification is issued to the user.			
Government Response	Agreed. SMEs provided language responding to	this comment.		



Paragraph

Post AWG

Proposed

Redlines

Newest

Redlines

WAS

IS200-1788	
Bits 78 through 81 of Message Type 40 shall provide the assumed Constellation Fault Probability (P_{const}) value for ARAIM at the current time for the associated GNSS constellation	
Bits 7882 through 8185 of Message Type 40 shall provide the assumed Constellation constellation Fault fault Probability rate (Pconst \underline{R}_{const}) value for ARAIM at the current time for the associated GNSS constellation.	

Bits 7882 through 8185 of Message Type 40 shall provide the assumed Constellation constellation <u>Fault fault</u> <u>Probability rate</u> (<u>PconstR</u>_{const}) value for ARAIM at the current time for the associated Proposed GNSS constellation.

> <u>R_{const} is the onset rate at which the instantaneous URE of two or more satellites exceed, due to a</u> common cause, 4.42 times the IAURA (or 5.73 times the IAURA when the conveying signal is provided with an enhanced level of integrity assurance).

> > APPROVED FOR PUBLIC RELEASE 36

Define R_{const}



Define R_{sat} and MFD_{sat}

STATEMS COMMAN						
DOORS ID	IS200-1797, IS200-2127, IS705-1644, IS705-174	I, IS800-1058, IS800-10	64			
Paragraph	IS-GPS-200 30.3.3.10.1.7 IS-GPS-200 30.3.3.10.1.8	Comment Number	CRM #85			
	Substantive	Disposition	Accept			
Comment Originator(s)	Bert Hayden (Aerospace)					
Comment	While providing responses to Europa's comment and MFDconst, Aerospace provided definitions Completeness of parameter definitions.) to clarify/define Rconst			
Government Response	Agreed. SMEs provided language responding be	yond Europa's commen	ts.			

of integrity assurance).



IS200-1797
Bits 74 through 77 of Message Type 40 shall provide the assumed Satellite Fault Rate (R_{sat}) value for ARAIM at the current time for the associated GNSS constellation.
Bits 74 through 77 of Message Type 40 shall provide the assumed <u>Satellitesatellite</u> Faultfault Raterate (Rsat) value for ARAIM at the current time for the associated GNSS constellation.
Bits 74 through 77 of Message Type 40 shall provide the assumed <u>Satellitesatellite</u> <u>Faultfault</u> <u>Raterate</u> (Rsat) value for ARAIM at the current time for the associated GNSS constellation. <u>R_{sat} is the onset rate at which the instantaneous URE of any given satellite exceeds 4.42 times the</u> IAURA (or 5.73 times the IAURA when the conveying signal is provided with an enhanced level
_

APPROVED FOR PUBLIC RELEASE 38

Define R_{sat}



Define MFD_{sat}

SYSTEMS COMMAND	
Paragraph	IS200-2127
WAS	<inserted object=""></inserted>
Post AWG Proposed Redlines	Bits 78 through 81 of Message Type 40 shall provide the assumed mean duration of a satellite fault (MFDsat) value for ARAIM at the current time for the associated GNSS constellation
Newest Proposed Redlines	Bits 78 through 81 of Message Type 40 shall provide the assumed mean duration of a satellite fault (MFDsat) value for ARAIM at the current time for the associated GNSS constellation. <u>MFD_{sat} is the mean time the instantaneous URE of any given satellite exceeds 4.42 times the IAURA (or 5.73 times the IAURA when the conveying signal is provided with an enhanced level of integrity assurance) without a timely notification issued to the user.</u>



CIEMS COMM								
	IS200-618, IS200-1371							
Paragraph	IS-GPS-200 20.3.3.5.1.1 IS-GPS-200 40.3.3.5.1.1	Comment Number	Number CRM #48					
Comment Type	Substantive	bstantive Disposition Accept With Comm						
Comment Originator(s)	D. Bouvet (Thales)							
Comment	During 2021 public ICWG, the following action was taken:							
	"The government will propose amendments to [IS-GPS-200] 20.3.3.5.1.1 and 40.3.3.5.1.1 to be reviewed at the 2022 Public ICWG that will specify the future use of the other Data IDs in LNAV transmissions. This proposal will preserve backward compatibility with all legacy receivers, whether or not they check the Data ID value."							
	No change detected in IS-GPS-200N Section 20.3.3.5.1.1. Consider updating the paragraph according to the action taken last year.							
Government Response	This is the same as PRAT Item 2021-05 and will This Special Topic presentation may or may not subject.	2						



Specific Alarm Condition c)

SYSTEMS COMMAND						
	IS200-1760					
Paragraph	IS-GPS-200 6.4.6.2.2	Comment Number	CRM #49			
Comment Type	Substantive	А	Accept with Comments			
Comment Originator(s)	D. Bouvet (Thales)					
Comment	Specific Alarm Condition c) says "The transmitted bits in words 3-10 in subframe 1, 2, or 3 are all set to 0's or all set to 1's."					
	In the case of "all set to 1's", please clarify whe impacted subframe are also set to 1, or whethe the case for the default navigation data)?	-				
Government Response	Clarification wording found for this item and a	oplied to the appropria	te paragraph			

SPACE SYSTEMS COMMAND	

Specific Alarm Condition c)

SYSTEMS COMMAND	
Paragraph	IS200-1760
Proposed Redlines	 C/A-Code or P(Y)-Code Signal (a) The failure of parity on 5 successive words of LNAV data (3 seconds) (see paragraphs 20.3.5 and 40.3.5). <i>(See Note 5)</i> (b) The broadcast IODE does not match the 8 LSBs of the broadcast IODC (excluding normal data set cutovers, see paragraph 20.3.3.4.1). (c) Bits 61 through 298 The transmitted bits in words 3-10 in subframe 1, 2, or 3 are all set to 0's or all set to 1's. (d) Default LNAV data is being transmitted in subframes 1, 2, or 3 (see paragraph 20.3.2). (e) The 8-bit preamble does not equal 10001011₂, decimal 139, or hexadecimal 8B (see paragraph 20.3.3).



MT40 content only applicable GNSS ID For GPS?

SYSTEMS COMMAND						
DOORS ID	IS705-1611					
Paragraph	IS-GPS-705 20.3.3.10	Comment Number CRM #50				
Comment Type	Substantive	Disposition	Accept			
Comment Originator(s)	D. Bouvet (Thales)					
Comment	The content of the ISM (MT40) seems appropriate to disseminate the Integrity Support Data of the GPS constellation, but not the ISD of the other constellations. A few examples:					
	 Proposed values for Rsat and MFDsat do not allow to encode the Psat of 3x10⁻⁵ for the GAL constellation (the closest value would be 3.16x10-5); Galileo will need to broadcast URA values through the ISM (not possible with the current format of MT40). 					
	It is suggested to clarify that the description of ID is equal to "0100" (GPS). For any other GNSS the message starting from bit 43 up to bit 276	ID, manufacturers sho				
Government Response	We agree the proposed text is correct.					



MT40 content only applicable GNSS ID For GPS?

Paragraph	IS200-1611 20.3.3.10
Proposed Redlines	 Figure 20-14a contains the structure of Message Type 40, Integrity Support Message (ISM) when the GNSS ID parameter is equal to "0100". When the GNSS ID parameter is different, bits 43 to 276 are reserved. The contents of Message Type 40 when the GNSS ID parameter is equal to "0100" are defined below, followed by material pertinent to the use of the ISM data. Users who implement Advanced Receiver Autonomous Integrity Monitoring (ARAIM) may use these parameters for the ARAIM algorithm as referenced in future TSO and MSO.



Simplify Table 20-XIc PRN Inclusion Mask Mapping

DOORS ID	IS705-1663		
Paragraph	IS-GPS-705 Table 20-XIc	Comment Number	CRM #51
Comment Type	Substantive	Disposition	Accept with Comments
Comment Originator(s)	D. Bouvet (Thales)		
Comment	For the PRN Mask, if it is accepted to limit the suggested to simplify Table 20-XIc "PRN Inclusion and deleting the other ones (MT40 format for or PRN mask).	on Mask Mapping" by ke	eping the GPS column only
Government Response	Decision is to keep GPS and SBAS columns Also affect IS-GPS-200 Table 30-XIc PRN Mappir Also affect IS-GPS-800 Table 3.5-11 PRN Mappir	5	



IS-GPS-705 Table 20-XIc

Paragraph	IS705-1663	}		Table	20-XIc P	'RN Map	ning				
				Tuble	20 710 1		ping			_	
Proposed		Bits	Galileo	GLONASS	BeiDou	GPS	SBAS	QZSS	IRNSS		
Redlines		<u>8993</u>	SVID 1	Freq. 1	RCN 1	PRN 1	PRN 120	PRN 183	PRN-ID-1	\leq	Keeping SBAS as
		<u>90</u> 94	SVID 2	Freq. 2	RCN-2	PRN 2	PRN 121	PRN 184	PRN ID-2		well.
		91 95	SVID 3	Freq. 3	RCN 3	PRN 3	PRN 122	PRN 185	PRN ID-3	(
		92 96	SVID-4	Freq. 4	RCN 4	PRN 4	PRN 123	PRN 186	PRN ID-4		
		<u>9397</u>	SVID-5	Freq. 5	RCN 5	PRN 5	PRN 124	PRN 187	PRN ID-5		
		<u>9498</u>	SVID 6	Freq. 6	RCN 6	PRN 6	PRN 125	PRN 188	PRN ID-6		
		95 99	SVID 7	Freq. 7	RCN 7	PRN 7	PRN 126	PRN 189	PRN ID-7		
		96 100	SVID 8	Freq. 8	RCN 8	PRN 8	PRN 127	PRN 190	Reserved		
		97 101	SVID 9	Freq. 9	RCN 9	PRN 9	PRN 128	PRN 191	Reserved		
		98 102	SVID 10	Freq. 10	RCN 10	PRN 10	PRN 129	PRN 192	Reserved		
		99 103	SVID 11	Freq. 11	RCN 11	PRN 11	PRN 130	PRN 193	Reserved		
		100 104	SVID 12	Freq. 12	RCN 12	PRN 12	PRN 131	PRN 194	Reserved		
		101 105	SVID 13	Freq. 13	RCN 13	PRN 13	PRN 132	PRN 195	Reserved		
		102 106	SVID 14	Freq. 14	RCN 14	PRN 14	PRN 133	PRN 196	Reserved		
		103 107	SVID-15	Freq. 15	RCN 15	PRN 15	PRN 134	PRN 197	Reserved		
		104 108	SVID 16	Freq. 16	RCN 16	PRN 16	PRN 135	PRN 198	Reserved	1	
		105 109	SVID 17	Freq. 17	RCN 17	PRN 17	PRN 136	PRN 199	Reserved	1	
		106 110	SVID 18	Freq. 18	RCN 18	PRN 18	PRN 137	PRN 200	Reserved	1	
		107 111	SVID 19	Freq. 19	RCN 19	PRN 19	PRN 138	PRN 201	Reserved		
		108 112	SVID 20	Freq. 20	RCN 20	PRN 20	PRN 139	PRN 202	Reserved	1	



b_nom and gamma_nom Clarification

SYSTEMS COMMAND					
	IS705-1649, IS705-1650, IS705-1653, IS200-1802, IS200-1803, IS200-1806, IS800-1051, IS800-1052, IS800-1055				
Paragraph	IS-GPS-705 20.3.3.10.1.5, 20.3.3.10.1.6 IS-GPS-200 30.3.3.10.1.5, 30.3.3.10.1.6 IS-GPS-800 3.5.4.7.1.5, 3.5.4.7.1.6	Comment Number	CRM #52		
Comment Type	Substantive	Disposition	Accept		
Comment Originator(s)	D. Bouvet (Thales)				
Comment	For the ARAIM, the Integrity Support Data (ISD) can include a nominal bias b_nom per satellite. MT40 transmits two values: b_nom and gamma_nom.				
	Clarify whether the transmitted b_nom corresponds to the ISD b_nom, or whether the receiver has to process the transmitted b_nom and the gamma_nom parameters to obtain the ISD b_nom value.				
	1. If the first option is the correct one, consider explaining the foreseen use of gamma_nom in the ARAIM in section 20.3.3.10.1.6.				
	2. If the second option is the correct one, consider adding the equation(s) to compute the ISD b_nom from the transmitted b_nom and gamma_nom parameters.				
Government Response	 The second option and we are adding a formula Clarification has been added to the subject para This response is modified by CRM #62 which has The resolution here also applies to corresponding 	agraphs as well (see next resulted in the term b _{nor}	slide) m being changed to B _{nom} 47		



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Sample Change for b_{nom} Clarification

Paragraph(s) of Interest	IS705-1650, IS705-1653
Proposed Redlines	20.3.3.10.1.5 <u>IAURA-Independent</u> Additive Term for Nominal Pseudorange Error Bias
	Bits 66 through 69 of Message Type 40 shall provide the assumed <u>IAURA-Independent</u> Additive Term ($\frac{\beta\beta_{nom}}{\beta_{nom}}$) value for ARAIM at the current time for the associated GNSS constellation the GPS
	satellites indicated in the PRN Inclusion Mask. The IAURA-Independent Additive Term(β_{nom})
	bounds additive biases in the instantaneous URE that do not scale with IAURA, which is defined in
	section 30.3.3.1.1.
	3.3.3.10.1.6 Scalar Term for Nominal Pseudorange Error Bias
	Bits 70 through 73 of Message Type 40 shall provide the assumed Scalar Term(γ_{nom}) value for
	ARAIM at the current time for the associated GNSS constellation the GPS satellites indicated in the
	PRN Inclusion Mask. The Scalar Term (γ_{nom}) bounds normalized additive biases in the instantaneous
	URE that scale with IAURA, which is defined in section 30.3.3.1.1.



Include RTCA/DO-246E Protocol

5200-1665 5-GPS-705 20.3.3.10.1.14 (Corrected) dministrative	Comment Number Disposition	CRM #53
dministrative		
	Disposition	
Rouwet (Thales)		Accept With Comments
. Douver (Thates)		
eferring to a proprietary document (RTCA/DO ne ISM CRC may be a concern for end users.	-246E change 1) to def	ine the protocol to process
onsider replacing this reference by a "free of PS-705.	charge" document, or	include the protocol in IS-
•		-
	e ISM CRC may be a concern for end users. nsider replacing this reference by a "free of S-705. cided to go a different direction. ther than include the protocol from DO-246E ue, we are going to remove the reference to	ferring to a proprietary document (RTCA/DO-246E change 1) to def e ISM CRC may be a concern for end users. nsider replacing this reference by a "free of charge" document, or i S-705. cided to go a different direction. ther than include the protocol from DO-246E in our documentation ue, we are going to remove the reference to this document in this



RTCA/DO-246E Protocol Decision

- m3 Low	
Paragraph	IS-GPS-705 20.3.3.10.1.14
WAS	Bits 245 through 276 of MT-40 are a 32-bit Cyclic Redundancy Check (CRC) specific to the ISM parameters. The ISM CRC will cover only the ISM parameters in Message Type 40, (Bits 39 to 244). Refer to DO-246E-Change 1 document for more details on the ISM CRC.
Public-ICWG Proposed IS	Bits 245 through 276 of MT-Message Type 40 are a 32-bit Cyclic Redundancy Check (CRC) specific to the ISM parameters. The ISM CRC will cover only the ISM parameters in Message Type 40, (Bbits 39 tothrough 244). Refer to DO-246E-Change 1 document for more details on the ISM CRC.
Current Proposed IS	Bits 245 through 276 of MT-Message Type 40 are a 32-bit Cyclic Redundancy Check (CRC) specific to the ISM parameters. The ISM CRC will cover only the ISM parameters in Message Type 40, (Bbits 39 tothrough 244). Refer to DO-246E-Change 1 document for more details on the ISM CRC.



Government Comments/Dispositions

CRM – COMBINED STAKEHOLDER/DIRECTORATE REVIEW STATUS:						
Disposition/Type	Critical	Substantial	Administrative	Totals	Concurrence	Notes
Accept		26	1	27		
Accept with Comment						
Reject		3		3		Government Comments FAA
Deferred		1		1		
Grand Totals:		30	1	31		

SPACE DISTERS COMMAND

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Limitations Of ASCII Characters For Civil Text Message

DOORS ID	IS200-2004, IS200-2115, IS800-282, IS800-1168	3	
Paragraph	IS-GPS-200 30.3.3.9 IS-GPS-800 3.5.4.5	Comment Number	CRM #54
Comment Type	Substantive (FAA)	Disposition	Accept
Comment Originator(s)	Jed Dennis		
Comment	Limitations of ASCII characters for civil text musefulness of the message.	essage is unnecessary a	nd potentially limits
Government Response	Agreed. Matching change to IS-GPS-800 has be	een added	



Message Types 36 and 15 Text Messages

Paragraph IS200-2004, IS200-2115

Newly Proposed Redlines

30.3.3.9 Message Types 36 and 15 Text Messages

Text messages are provided either in Message Type 36, Figure 30-9, or type 15, Figure 30-14. The specific contents of text message will be at the discretion of the Operating Command. Message Type 36 can accommodate the transmission of 18 eight-bit ASCII ISO 8859-1 characters. Message Type 15 can accommodate the transmission of 29 eight-bit ASCII ISO 8859-1 characters. The requisite bits shall occupy bits 39 through 274 of Message Type 15 and bits 128 through 275 of Message Type 36. Messages that include control and other undefined characters are for special uses. Such messages are valid, but may appear garbled if displayed.

The eight-bit ASCII characters shall be limited to the set described in paragraph 20.3.3.5.1.8.



CM-Code Signal Clarification

SYSTEMS COMMAND			
DOORS ID	IS200-1762		
Paragraph	IS-GPS-200 6.4.6.2.2	Comment Number	CRM #55
Comment Type	Substantive	Disposition	Accept
Comment Originator(s)	Jed Dennis (FAA)		
Comment	For CM-code signal, clarify that any of the MT- alert. As written, it requires all of these to be (d) The transmitted bits (bits 39-276) is and/or 30's are all set to 0's or all s	all 0's or all 1's to be c n <u>one or more of Message</u>	onsidered an alert.
Government Response	Agreed		



Additional Guidance On Message Type 0

and Luna			
DOORS ID	IS200-1762		
Paragraph	IS-GPS-200 6.4.6.3	Comment Number	CRM #56
Comment Type	Substantive	Disposition	Accept
Comment Originator(s)	Jed Dennis (FAA)		
Comment	Is there additional guidance on how to apply this marginal criteria? While receiver might be able to confirm a MT-0, it is difficult to determine that it replaced one of these message types, especially if the receiver fails to correctly decode some messages. Are there other cases were an MT-0 might replace a MT-31 to MT-39 due to lack of information other than clock information?		
Government Response	Agreed		
4			



Clarification on 6.4.6.3 Marginal Indications

SYSTEMS COMMAND	
Paragraph	IS200-1762
Newly Proposed Redlines	The health of the CM-code and CL-code signals is marginal when the signals would otherwise have been defined as healthy except that one or more of the following three warning conditions is or are present:
	 Default CNAV data (i.e., Message Type 0) is being transmitted in lieu of Message Type 10, 11 and/or Message Type 30's on the CM-code signal (e.g., a current and consistent CEI data set is not available within the maximum broadcast interval defined in paragraph 30.3.4.1). See paragraph 30.3.3. The satellite does not broadcast a current and consistent CEI data set within three times the maximum broadcast interval defined in paragraph 30.3.4.1. (Notes 1 and 2)
	2. The URA alert flag is raised (i.e., bit 38 of each CNAV message is set to 1) and therefore the CM-code signal URA components do not apply to the CM-code and CL-code signals. This means the CM-code and CL-code signal URA may be worse than indicated by the URA index components transmitted in Message Type 10 and Message Type 30's. See paragraph 30.3.3.
	3. Either or both the URAED index in Message Type 10 and the URANED0 index in Message Type 30's transmitted in the CM-code signal are equal to 15 or -16 ("N"=15 or "N"=-16). See paragraphs 30.3.3.1.1.4 and 30.3.3.2.4.
	Note 1: Default CNAV data (i.e. Message Type 0) may be transmitted in lieu of any message type when the correct data for the message type is unavailable or when no other message is scheduled.
	Note 2: UE might be unable to confirm the satellite broadcast of a consistent data set when local conditions prevent correctly receiving and decoding a continuous set of messages.



Remove PRN37 As An Alert Mechanism

SYSTEMS COMMAND			
DOORS ID	IS200-1335		
Paragraph	IS-GPS-200 6.4.3	Comment Number	CRM #57
Comment Type	Administrative	Disposition	Accept
Comment Originator(s)	Jed Dennis (FAA)		
Comment	 Should remove PRN37 as an alert mechanism from Section 6.4.6. Also, if PRN37 is not being used although this is not critical. 6.4.3 PRNs 33 and 37 PRN 33 should not be used by satellite ground applications. PRN 37 should not be used by satellite and the statement of t	for SATZAP, then shoul s because of its prior u ot be used by satellites	d remove reference in IS, se in specialized
Government Response	Agreed		57
1			



APPROVED FOR PUBLIC RELEASE Add Wn_{op} to Table 6-I-1 CEI Data Set Parameters

SYSTEMS COMMAND			
DOORS ID	IS200-1639		
Paragraph	IS-GPS-200 Table 6-I-1	Comment Number	CRM #58
Comment Type	Substantive	Disposition	Accept
Comment Originator(s)	Jed Dennis (FAA)		
Comment	Table should include WNop. WNop is required	in the calculation of IA	URA.
Government Response	Agreed		



Table 6-I-1

Message N/A N/A 10 30 30-37 30-37 30-37 30-37 30-37 N/A

Paragraph	IS200-1639			
Newly		Symph ol	Parameter Name	Subframe
Proposed Redlines		Symbol SV Health		
Neutines		SV Health	SV Health (6 bits)	1
		IODC	Issue of Data, Clock	1
		URA	URA Index	1
		WN	Data Sequence Propagation Week Number	1
		T _{GD}	Group Delay Differential	1
		a _{f0}	SV Clock Bias Correction Coefficient	1
		a _{fl}	SV Clock Drift Correction Coefficient	1
		a _{f2}	Drift Rate Correction Coefficient	1
		t _{oc}	Time of Clock	1
		<u>WN_{op}</u>	CEI Data Sequence Propagation Week Number	<u>N/A</u>
		\sqrt{A}	Square Root of the Semi-Major Axis	2
		•		
		•		
		·	1	l,

SPACE

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Type 30 Rate

SYSTEMS COMMAND			
DOORS ID	IS200-670		
Paragraph	IS-GPS-200 Table 30-XII	Comment Number	CRM #59
Comment Type	Substantive	Disposition	Defer
Comment Originator(s)	Jed Dennis (FAA)		
Comment	Type 30 should be at the same rate as Type 10, and WN _{op}	11, as it has critical pa	arameters like T _{gd} , ISC,
Government Response	This subject is related to Bert Hayden's Special Recommendation". It is advisable to defer this the context of that Special Topic later in the dates are special topic later in topi	s comment for the mon	



Specific Alarm Indications Clarification

OTSTEMS COMMAND				
DOORS ID	IS200-1760			
Paragraph	IS-GPS-200 6.4.6.2.2 Specific Alarm Indications	Comment Number	CRM #60	
Comment Type	Substantive	Disposition	Accept	
Comment Originator(s)	Hamza Abduselam (FAA)			
Comment	Add a note to clarify that default LNAV data broadcast in Subframe 4 or in Subframe 5 does not constitute an alarm condition.			
Government Response	Agree. This added text is an extension to Not	e 5 in this paragraph		



6.4.6.2.2 Specific Alarm Indications

atsteas continues	
Paragraph IS200-1760	
 Newly Proposed Redlines 5. An alarm indication (see C/A-Code or P(Y)-Code Signal (a)) does not apply to the defadescribed in paragraph 20.3.2, when in subframes 4 or 5. Application of the user parity paragraph 20.3.5.2 will result in failed parity checks for words 3-10 because the defaut pattern is applied to bits 61-298. According to a) and d) default LNAV data broadcast subframe 5 will not be considered as a do-not-use condition, and the user equipment in the GPS L1 measurement as healthy so long as none of the other conditions leading to UNHEALTHY determination are present. 	y algorithm at lt LNAV data <u>in subframe 4 or in</u> nay continue using

Change b_{nom_0} Value Range

SPACE SYSTEMS COMMAND			
DOORS ID	IS200-1804, IS705-1651, IS800-1053		
Paragraph	IS-GPS-200 30.3.3.10.1.5 IS-GPS-705 20.3.3.10.1.5 IS-GPS-800 3.5.4.7.1.5	Comment Number	CRM #61, 69, 77
Comment Type	Substantive	Disposition	Accept
Comment Originator(s)	Jed Dennis (FAA)		
Comment	Need to review Bnom,0 value ranges. Bnom,0 will be added with Ynom*URA to achieve actual Bnom for the ARAIM algorithm. Current ranges seem higher than needed and approach value of the URA.		
	Preliminary analysis reported by Stanford on current LNAV performance indicates a bias of about 0.15*URA, or roughly 0.36-m attributed to satellite position error bias. Analysis of signal deformation, code carrier covergence and other sources of bias has resulted in consideration of bias values in the 0.75 to 1.0-m range. Given that this directly inflates the protection level and is combined with gamma_nom * URA, it seems better to enable lower ranges in the messages.		
Government	Agreed.		
Response	b_{nom} is now B_{nom} per CRM #62 (see below		
1		APE	PROVED FOR PUBLIC RELEASE 63



30.3.3.10.1.5 Additive Term for Nominal Pseudorange Error Bias

SYSTEMS COMMAND		
Paragraph	IS200-1760	
	WAS	Newly Proposed
	The four bits are defined as follows:	The four bits are defined as follows:
	0000 = 0.00 meters	0000 = 0.00 meters
	0001 = 0.13 meters	$\overline{0001 = 0.10 \text{ meters}}$
	0010 = 0.25 meters	$\overline{0010 = 0.20 \text{ meters}}$
	0011 = 0.38 meters	$\overline{0011 = 0.30 \text{ meters}}$
	0100 = 0.50 meters	$\overline{0100 = 0.40 \text{ meters}}$
	0101 = 0.63 meters	$\overline{0101 = 0.50 \text{ meters}}$
	0110 = 0.75 meters	$\overline{0110 = 0.60 \text{ meters}}$
	0111 = 0.88 meters	$\overline{0111 = 0.70 \text{ meters}}$
	1000 = 1.00 meter	1000 = 0.80 meters
	1001 = 1.13 meters	$\overline{1001 = 0.90 \text{ meters}}$
	1010 = 1.25 meters	1010 = 1.00 meter
	1011 = 1.38 meters	$\overline{1011} = 1.10 \text{ meters}$
	$\frac{1100 = 1.50 \text{ meters}}{1100 = 1.50 \text{ meters}}$	1100 = 1.20 meters
	1101 = 1.63 meters	1101 = 1.30 meters
	1110 = 1.75 meters	$\overline{1110 = 1.40 \text{ meters}}$
	$\frac{1111 - 2.00 \text{ meters}}{1111 - 2.00 \text{ meters}}$	1111 = 1.50 meters 64



Nomenclature For B_{nom_0}

SYSTEMS COMMAND					
DOORS ID	IS200-1770, IS200-1803, IS200-1808, IS705-1618	8, IS705-1606, IS800-10	40, IS800-1030		
Paragraph	IS-GPS-200 Table 30-XIa - ISM Parameters, 30.3.3.10.1.5, Figure 30-14a IS-GPS-705 Table 20-XIa - ISM Parameters, Figure 20-14a. Message Type 40 -ISM IS-GPS-800 Table 3.5-9 ISM Parameters, Figure 3.5-8a Subframe 3, Page 8, ISM	Comment Number	CRM #62, 63, 70, 71, 78, 79		
Comment Type	Substantive	Disposition	Accept with Comments		
Comment Originator(s)	Jed Dennis (FAA)				
Comment	62, 70, 78 Nomenclature for bnom,0 does not match planned nomenclature for MOPS and could cause confusion.				
	(Later corrected to Update nomenclature to B_{nom_0})				
	(Even later b0)?				
	63 Update nomenclature for Bnom,0 in the Table 3	80-XIa - ISM Parameters			
	71 Update nomenclature for Bnom,0 in the Table 2	0-XIa - ISM Parameters			
	79 Update nomenclature for Bnom,0 in the Table 3	3.5-9 ISM Parameters			
Government Response	Agreement to change b_{nom} to B_{nom} (beta sub nom). Ea	ach document has various	figures and tables to update. $_{65}$		



Table 30-XIa - ISM Parameters

CONTRACT CONTRACT							
Paragraph	IS200-1760						
Newly Proposed Redlines		Parameter GNSS ID	No. of Bits* <u>*</u> 4	Scale Factor (LSB)	Valid Range** <u>*</u> <u>See text</u>	Units	
		WN _{ISM}	13	1	0 to 8191	weeks	
		TOW _{ISM}	6	4	0 to 164	hours	
		t _{correl}	4		0 to 12 See text	hours	
		<mark>b</mark> βnom	4		0 to 2 See text	meters	
		γ _{nom}	4		0 to 2 See text		
Name		R _{sat}	4	1x10⁻²	³ -t o 3.16x10⁻¹⁰ <u>See tex</u>	t /hours	
change is	_	<u>MFD</u> _{sat}	<u>4</u>		See text		
here		PRconst	<u>4</u>		See text		
		MFD _{const}	4	0.	.25 to 2 4 <u>See text</u>	hours	
		Service Level* <u>**</u>	3		See text		
		PRN Inclusion Mask ****	63		See text		
		* See Figure 30-14a for	•		• •		
		** Unless otherwise indic indicated bit allocation			is the maximum rang	ge attainable with	
		*** See Table 30-XIb for Service Level Descriptions					
		**** See Table 30-XIbc for		•	ng		
							<u> </u>



review ^ynom value ranges

SYSTEMS COMMAND				
DOORS ID	IS200-1807, IS705-1654, IS800-1056			
Paragraph	IS-GPS-200 30.3.3.10.1.6 IS-GPS-705 20.3.3.10.1.6 IS-GPS-800 3.5.4.7.1.6	Comment Number	CRM #64, 72, 80	
Comment Type	Substantive	Disposition	Accept	
Comment Originator(s)	Jed Dennis (FAA)			
Comment	Need to review Ynom value ranges. Current range appears too high and only one or two values would ever be used. Ynom should be a small percentage of URA. Preliminary analysis of data seems to suggests Ynom values of 0.1, 0.15 or 0.2 are realistic.			
	With a URA of 2.4, this results in a Bnom contribution of 0.36 for gamma_nom = 0.15. With a URA of 1.0, this can still be achieved with gamma_nom = 0.40. There appears to be margin for low URAs and provides better tuning if adjusted for large URAs.			
	Better align parameter range with observation	S.		
Government Response	Agreed			

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30.3.3.10.1.5 Additive Term for Nominal Pseudorange Error Bias

Paragraph IS200-1807 WAS Newly Proposed The four bits are defined as follows: The four bits are defined as follows	
The four bits are defined as follows: The four bits are defined as follows	
0000 = 0.00	
0001 = 0.13 $0001 = 0.05$	
0010 = 0.25 $0010 = 0.10$	
0011 = 0.38 $0011 = 0.15$	
0100 = 0.50 $0100 = 0.20$	
0101 = 0.63 $0101 = 0.25$	
0110 = 0.75 $0110 = 0.30$	
0111 = 0.88 $0111 = 0.35$	
$1000 = 1.00$ $\overline{1000 = 0.40}$	
1001 = 1.13 $1001 = 0.45$	
$1010 = 1.25$ $\overline{1010 = 0.50}$	
1011 = 1.38 $1011 = 0.55$	
1100 = 1.50 $1100 = 0.60$	
$\overline{1101 - 1.63}$ $\overline{1101 = 0.65}$	
$\overline{1110 = 1.75}$ $\overline{1110 = 0.70}$	
1111 = 2.00 $1111 = 0.75$	68



Add section on protocols for $\mathrm{B}_{\mathrm{nom}}$

SYSTEMS COMMAND			
DOORS ID	After IS200-1765		
Paragraph	IS-GPS-200 30.3.3.10.2 IS-GPS-705 20.3.3.10.2 IS-GPS-800 3.5.4.7.2	Comment Number	CRM #65, 66, 67, 73, 74, 75, 81, 82, 83
Comment Type	Substantive	Disposition	Accept
Comment Originator(s)	Jed Dennis (FAA)		
Comment	Add section on protocols		
	Add paragraph on calculation of Bnom		
	Add equation for calculation of Bnom		
Government Response	b _{nom} is now β _{nom} per CRM #62 (see above)		



Paragraph

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Proposed 30.3.3.10.2 Use of ISM Data

(Insert after IS200-1765)

Newly Proposed Insertion

30.3.3.10.2 Use of ISM Data

To calculate the nominal pseudorange error bias (b_{nom}), use the following equation

 $\underline{\mathbf{b}}_{\underline{\mathrm{nom}}} = \underline{\beta}_{\underline{\mathrm{nom}}} + \underline{\gamma}_{\underline{\mathrm{nom}}} \underline{\mathrm{IAURA}}$

IAURA is defined in section 30.3.3.1.1.



Satellite Fault Rate Values Range Too Extensive?

IS200-1798, IS705-1645, IS800-1059			
IS-GPS-200 30.3.3.10.1.7 IS-GPS-705 20.3.3.10.1.7 IS-GPS-800 3.5.4.7.1.7	Comment Number	CRM #68, 76, 84	
Substantive	Disposition	Reject	
Jed Dennis (FAA)			
There has been no discussion to provision equipment with the ability to operate with the higher-integrity URA that is included in the IS. Therefore these values will always need to be set for the current 4.42 integrity setting.			
The argument is to leave as-is because the bit encod systems	ing applies to the interfac	ce and not just current GNSS	
	IS-GPS-200 30.3.3.10.1.7 IS-GPS-705 20.3.3.10.1.7 IS-GPS-800 3.5.4.7.1.7 Substantive Jed Dennis (FAA) Range of values for Satellite Fault Rate seems to ext constellation's SARPs commitments (GLONASS 1e ⁻⁴). result in ARAIM operating without checking single sat at least should set the very optimistic values to "rese continue to check single-fault modes, which could be or Code-Noise multipath. There has been no discussion to provision equipment URA that is included in the IS. Therefore these value integrity setting. Rationale: Safety The argument is to leave as-is because the bit encod	IS-GPS-20030.3.3.10.1.7Comment NumberIS-GPS-70520.3.3.10.1.7IS-GPS-8003.5.4.7.1.7SubstantiveDispositionJed Dennis (FAA)DispositionRange of values for Satellite Fault Rate seems to extensive. The large values constellation's SARPs commitments (GLONASS 1e-4). The low values, with dyna result in ARAIM operating without checking single satellite faults. While we cat least should set the very optimistic values to "reserved" so that current gen continue to check single-fault modes, which could be satellite-induced or could or Code-Noise multipath.There has been no discussion to provision equipment with the ability to operation upwatches and the IS. Therefore these values will always need to be sintegrity setting.Rationale: SafetyThe argument is to leave as-is because the bit encoding applies to the interface	



30.3.3.10.1.7 Satellite Fault Rate

Paragraph	IS200-1807		
	WAS	Newly Proposed (but Rejected)	
	The four bits are defined as follows:	The four bits are defined as follows:	
	0000 = 0.00 meters	0000 = 0.00 meters	
	0001 = 0.13 meters	$\overline{0001 = 0.10 \text{ meters}}$	
	0010 = 0.25 meters	$\overline{0010 = 0.20 \text{ meters}}$	
	0011 = 0.38 meters	$\overline{0011 = 0.30 \text{ meters}}$	
	0100 = 0.50 meters	$\overline{0100 = 0.40 \text{ meters}}$	
	0101 = 0.63 meters	$\overline{0101 = 0.50 \text{ meters}}$	
	0110 = 0.75 meters	$\overline{0110 = 0.60 \text{ meters}}$	
	0111 = 0.88 meters	$\overline{0111 = 0.70 \text{ meters}}$	
	1000 = 1.00 meter	$\overline{1000 = 0.80 \text{ meter}}$	
	1001 = 1.13 meters	$\overline{1001} = 0.90 \text{ meters}$	
	1010 = 1.25 meters	$\overline{1010 = 1.00 \text{ meters}}$	
	1011 = 1.38 meters	$\overline{1011 = 1.10 \text{ meters}}$	
	1100 = 1.50 meters	1100 = 1.20 meters	
	$\frac{1101 - 1.63 \text{ meters}}{1101 - 1.63 \text{ meters}}$	1101 = 1.30 meters	
	1110 = 1.75 meters	$\overline{1110 = 1.40 \text{ meters}}$	
	1111 = 2.00 meters	1111 = 1.50 meters 72	



Open RFC-495 Discussion

QUESTIONS & COMMENTS?

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Action Item Review

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Global Positioning System (GPS) Public Interface Control Working Group (ICWG) & Public Forum Special Topic Briefings

October 26, 2022 0830-1600 PDT

United States Space Force Positioning, Navigation, and Timing Mission Area



Expected Almanac Behavior in OCX Era

October 26, 2022

Dr. Andrew Hansen, DOT Volpe Center



Background and Motivation

- GPS almanac reference time (t_{oa}) is defined in the respective Interface Specifications for LNAV, CNAV, CNAV-2 messages
- In all cases it is the count of seconds into the applicable GPS almanac reference week number (WN $_{\rm a})$
- Operationally, t_{oa} is managed by the control segment to both
 - occur approximately 70 hours after first valid Signal in Space (SIS) transmission and
 - ensure that the difference between any active t_{oa} and GPS time (t) is less than 3.5 days
- This operational flexibility with the IS compliant definition leaves room for different $t_{\rm oa}$ generation



Time of Almanac (t_{oa}) Behavior Comparison

Operational Assignment of Time of Almanac (t_{oa}) has flexibility

- AEP era control segment
 - Nominal almanac generation occurs daily at 2200 as a series of five fit intervals with the first fit starting at the 2200 epoch for the given day
 - Any off-nominal/contingent almanac generation also starts the first fit interval at the 2200 epoch of the given day even if that epoch is in the past
- OCX era control segment
 - Nominal almanac generation can occur once or twice a day as a series of five fit intervals
 - Off-nominal/contingent almanac generation produces five fit intervals
 - Regardless of which generation process occurs, the start time of the first fit interval is always the most recent zero-age-of-data epoch

There is Operational Flexibility in the Generation of Almanac Reference Time!

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Example Almanac Intervals and t_{oa}

Almanac generated on the 2200 epoch GPS time of Fit AEP toa AEP toa OCX toa Week AEP Fit start (tgs) secondsOfWeek OCX Fit start (tgs) secondsOfWeek predict Duration Week (weeks) (tgs) (weeks) (days) (sec)

Almanac NOT generated on the 2200 epoch

GPS time of predict (tgs)	Fit Duration (days)	AEP Fit start (tgs)	AEP toa Week (weeks)	AEP toa secondsOfWeek (sec)	OCX Fit start (tgs)	OCX toa Week (weeks)	OCX toa secondsOfWeek (sec)
984317400	6	984261600	1627	503808	984317400	1627	552960
984317400	6	984780000	1628	405504	984835800	1628	466944
984317400	6	985298400	1629	319488	985354200	1629	380928
984317400	32	985816800	1630	233472	985872600	1630	294912
984317400	32	988581600	1634	589824	988637400	1635	36864

 \mathbf{t}_{ps} units are total GPS seconds since the start of GPS time

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OCX toa

(sec)



Summary

- Both AEP and OCX produce IS-GPS-200/705/800 compliant almanacs
- Behavior between AEP and OCX will be slightly different from a user perspective
- After OCX becomes operational the users will see the following:
 - Nominally, \mathbf{t}_{oa} values produced by OCX will be the same as AEP
 - However, \mathbf{t}_{oa} will be as much as a day later than that from AEP for new uploads
 - And, when off-nominal/contingent almanacs are generated by OCX, the t_{oa} produced will be arbitrary with respect to the nominal almanacs (2200 epoch)

There is Operational Flexibility in the Generation of Almanac Reference Time!

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Action Item Review



LUNCH BREAK





Mitigate LNAV Default Nav Anomaly

October 26, 2022

Hamza Abdulselam Federal Aviation Administration

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Objective: Mitigate the major hazard posed by a possible repeat of the 22 July 2020 broadcast of default navigation data (DND) in Subframe 4 page 17.

Owner: Hamza Abdulselam (FAA) and Bert Hayden (Aerospace)

Progress/Accomplishments: Assessed impact to aviation and made the necessary modifications at the interface, control segment and user equipment levels

✓ Interface Specification

- ✓ Held Special topic discussion on anomaly hazard at the GPS Public ICWG (Sep 2021)
- ✓ Added clarifying language at the Public ICWG
- ✓ Control Segment
 - ✓ Installed AEP modification to prevent/eliminate 22 July 2020 root cause (April 2022)
 - Estimated frequency of subframes 4/5 DND for healthy, operational satellites and found no other case (Feb 2022)
 - Run a test during IST 3-1 Phase 1B, in Feb 2023, using the OCX simulators (TSTS) and develop a directive with expected data to replicate the event and assess the actual results and effects. ECD Feb 2023
- ✓<u>User Equipment</u>
 - ✓ Assessed impact on current avionics with OEMs though RTCA (May 2021)
 - ✓ Validated aviation requirements through consultation with RTCA (Aug 2021)
 - ✓ Updated language adapted in New draft MOPS (e.g. ED 259) (March 2022)
 - Language clarifies that DND in subframe 4 and 5 by itself is not an alarm condition Applies to future avionics
 - Note to Manufacturers posted next to TSO-C145e and TSO-C196b Advises manufacturers to apply the language adapted by the draft MOPS to their current design (May 2022)
 - This approach will only affect new avionics. Current thinking is to exempt default navigation data in Subframes 4 and 5 from the five successive parity failures test

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CNAV Message Schedules CSOC IPT Recommendation

October 26, 2022

Bert Hayden The Aerospace Corp. Karl Kovach The Aerospace Corp.

			APPRC	OVED F	OR PL	JBLIC I	RELEA	SE															_		
	*	IIR	-M	*	IIF		/ II	I/IIIF			Worl	ks fo	r Blo	ck	*	Not	Idea	al for	Bloc	k	[el	m	ola	at	e
mme	ende	d In	nitial S	Sche	dule	for G	iPS-I	/	-											-		· • • ſ			
MMAND																									
	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25
1	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10
2	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11
3	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30
4	37	37	37	37	37	37	37	37	37	37	37	37	37	37	37	37	37	37	37	37	37	37	37	33	35
sec	24	48	72	96	120	144	168	192	216	240	264	288	312	336	360	384	408	432	456	480	504	528	552	576	600
		Γ	Conte	ent				Curre	nt L5	Speci	ficatio	on	This S	ched	ule				* "(Intic	nal″	Mes	<u>с э п о</u>	c	
MT	10		Ephe	meris	1			24 se		24 sec								Ċ	Jptic	ла	IVICS	sage	3		
MT	11		Ephe	meris	2			24 sec 24 sec																	
MTS	3X		Clock Correction					24 sec					24 sec												
MTS	30		TGD/	ISC/Io	nospł	nere		144 s	ec				24 se	с				•							
MT 3	33		UTC					144 s	ec				600 s	ec					<u> р</u>	riarit	izoc	150 2	nd A	lman	
MTS	35		GGTC) *				144 s	ec				600 s	ec				•		IOIII	1283	ISC a	nu A	IIIaII	ac
MT 4	40		ISM *	¢				144 s	ec				N/A												
MT	32		EOP *	k				900 s	ec				600 s	ec											
MT	37		Midi	Almar	nac *			const	ellatio	on in 6	50 mir	n 🗌	126 P	RNs i	n 60 r	nin		•							
MT	13 / 1	.4	Differ	rential	Corre	ection	S *	const	ellatio	on in 1	L5 mir	า	N/A												
MT :	15		Text N	Vessa	ge *			none					600 s	ec											



🖌 III/IIIF

Template B

Recommended Initial Schedule for GPS-IIF

		1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25
	1	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10
	2	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11
ſ	3	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30
Ī	4	33	35	33	35	33	35	33	35	33	35	33	35	33	35	33	35	33	35	33	35	33	35	33	35	33
-	sec	24	48	72	96	120	144	168	192	216	240	264	288	312	336	360	384	408	432	456	480	504	528	552	576	600

	Content	Current L5 Specification	This Schedule
MT 10	Ephemeris 1	24 sec	24 sec
MT 11	Ephemeris 2	24 sec	24 sec
MT 3X	Clock Correction	24 sec	24 sec
MT 30	TGD/ISC/Ionosphere	144 sec	24 sec
MT 33	UTC	144 sec	96 sec
MT 35	GGTO *	144 sec	96 sec
MT 40	ISM *	144 sec	N/A
MT 32	EOP *	900 sec	96 sec
MT 37	Midi Almanac *	constellation in 60 min	N/A
MT 13 / 14	Differential Corrections *	constellation in 15 min	N/A
MT 15	Text Message *	none	600 sec

* "Optional" Messages

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IIR-M IIF

🖌 III/IIIF

Works for Block * Not Ideal for Block Template C

Recommended Initial Schedule for GPS-IIR-M

	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
1	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10
2	11	11	11	11	11	11	11	11	11	11	11	11	11	11	11
3	30	30	30	30	30	30	30	30	30	30	30	30	30	30	30
4	33	35	33	35	33	35	33	35	33	35	33	35	33	35	33
sec	24	48	72	96	120	144	168	192	216	240	264	288	312	336	360

	Content	Current L5 Specification	This Schedule
MT 10	Ephemeris 1	24 sec	24 sec
MT 11	Ephemeris 2	24 sec	24 sec
MT 3X	Clock Correction	24 sec	24 sec
MT 30	TGD/ISC/Ionosphere	144 sec	24 sec
MT 33	UTC	144 sec	96 sec
MT 35	GGTO *	144 sec	96 sec
MT 40	ISM *	144 sec	N/A
MT 32	EOP *	900 sec	192 sec
MT 37	Midi Almanac *	constellation in 60 min	N/A
MT 13 / 14	Differential Corrections *	constellation in 15 min	N/A
MT 15	Text Message *	none	600 sec

* "Optional" Messages

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ICD Recommendations

- Replace minimum broadcast intervals in public ICDs with general specifications
- Primary data (CEI) no less than every 30 seconds, 24 seconds nominal
- Secondary data every ten minutes from constellation (rather than a single SV)

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ISC and CEI Table

October 26, 2022

Karl Kovach The Aerospace Corp.

The Issue (1 of 3)

6.2.8 Clock, Ephemeris, Integrity (CEI) Data Set.

The Clock, Ephemeris, Integrity (CEI) data set is the collection of SV-specific clock correction polynomial parameters, ephemeris parameters, and related parameters (health flags, URA parameters, time tags, etc.) needed to use the SV's broadcast signal(s) in the positioning service. The parameters in the CEI data set are explicitly listed in Table 6-I-1. The entire CEI data set is needed for maximum accuracy. However, the core CEI data set (parameters without NOTE1 in Table 6-I-1) is sufficient for an initial position solution. The top term provides the epoch time of week of the state data utilized for the core CEI data set.

6.2.8.1 Core CEI Data Set.

A Core CEI Data Set are the CEI parameters necessary for a satellite to be used for a position solution (non-almanac); broadcast to users with the shortest broadcast interval -- see Table 20-XII. The top term provides the epoch time of week of the state data utilized for CEI data, except for parameters marked with a Note1 in Table 6-I-1.



The Issue (2 of 3)

Table 6-I-1. CEI Data Set Parameters

IS-GPS-705H 23 Mar 2021

Symbol	Parameter Name	Message
á	Change Rate in Semi-major Axis	10
ΔA	Semi-major Axis Difference at Reference Time	10
Δn_0	Mean Motion Difference from Computed Value at Reference Time	10
Δn_0	Rate of Mean Motion Difference from Computed Value	10
ω	Argument of Perigee	10
е	Eccentricity	10
ISF	Integrity Status Flag NOTE1	10
(L1/L2/L5)	Signal Health (3 bits)	10
Mo	Mean Anomaly at Reference Time	10
URAED	Elevation Dependent User Range Accuracy	10
WN	Week Number	10
toe	Time of Ephemeris	10, 11
top	CEI Data Sequence Propagation Time of Week	10, 30-37
Ω	Rate of Right Ascension	11
Ωο	Longitude of Ascending Node of Orbit Plane at Weekly Epoch	11
Cic	Amplitude of the Cosine Harmonic Correction Term to the Angle of Inclination	11
Cis	Amplitude of the Sine Harmonic Correction Term to the Angle of Inclination	11
Crc	Amplitude of the Cosine Harmonic Correction Term to the Orbit Radius	11
Crs	Amplitude of the Sine Correction Term to the Orbit Radius	11
Cuc	Amplitude of Cosine Harmonic Correction Term to the Argument of Latitude	11
Cus	Amplitude of Sine Harmonic Correction Term to the Argument of Latitude	11
io	Inclination Angle at Reference Time	11
io-n-DOT	Rate of Inclination Angle	11



The Issue (3 of 3)

		23 Mar 2021
ISCL1C/A	Inter-signal Correction	30
ISC _{L2C}	Inter-signal Correction	30
ISC _{L515}	Inter-signal Correction	30
ISC _{L505}	Inter-signal Correction	30
T _{GD}	Group Delay Differential	30
a _{fo}	SV Clock Bias Correction Coefficient	30-37
a _{f1}	SV Clock Drift Correction Coefficient	30-37
a _{f2}	Drift Rate Correction Coefficient Index	30-37
toc	Time of Clock	30-37
URANEDO	NED Accuracy Index	30-37
URA _{NED1}	NED Accuracy Change Index	30-37
URA _{NED2}	NED Accuracy Change Rate Index	30-37
Alert	Alert Flag NOTE1	All
curve fit. I Updates to par	arameters so indicated are for CEI Refinemer Parameters not indicated are needed for/limite ameters in table shall prompt changes in t_{oe}/t_{oc} . Any parameter n or without a change in t_{oe}/t_{oc} .	ed to curve fit.

IS-GPS-705H



Recommended Clarification

		IS-GPS- 23 Mar
ISCL1C/A	Inter-signal Correction NOTE1	30
ISC _{L2C}	Inter-signal Correction NOTE1	30
ISC _{L515}	Inter-signal Correction NOTE1	30
ISC _{L303}	Inter-signal Correction NOTE1	30
T _{GD}	Group Delay Differential NOTE1	30
afo	SV Clock Bias Correction Coefficient	30-37
a _{f1}	SV Clock Drift Correction Coefficient	30-37
a _{f2}	Drift Rate Correction Coefficient Index	30-37
toc	Time of Clock	30-37
URANEDO	NED Accuracy Index	30-37
URA _{NED1}	NED Accuracy Change Index	30-37
URA _{NED2}	NED Accuracy Change Rate Index	30-37
Alert	Alert Flag NOTE1	All
dicated are i	I neters so indicated are for CEI Refinement – not limited to curve fit. F needed for/limited to curve fit. rameters in table shall prompt changes in t _{oe} /t _{oc} . Any parameter man	
Ipdates to pa		rked with NOTE1

Should Be

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Observations

- Historically, T_{GD} updated quarterly
 - Clearly not limited to 3-hour CNAV curve fit
 - T_{GD} changes very slowly
- Currently, ISCs updated approximately semi-annually
 - Clearly not limited to 3-hour CNAV curve fit
 - ISCs change very slowly
- In OCX era, ISCs likely 'slightly updated' daily
 - Clearly not limited to 3-hour CNAV curve fit

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Data ID Issue

October 26, 2022

Bert Hayden The Aerospace Corp. Karl Kovach The Aerospace Corp.



20.1 Scope.

This appendix describes the specific legacy navigation (LNAV) data structure denoted by data ID number 2 for the lower set of PRN numbers (PRN 1-32). This data ID number, when transmitted as part of the LNAV data, shall be represented by the two-bit binary notation as 01. Data ID number 1 is no longer in use. The LNAV data structure for the lower set of PRN numbers is denoted as LNAV-L. The LNAV data structure for the upper set of PRN numbers (LNAV-U) is described in Appendix IV.

20.3.3.5.1.1 Data ID and SV ID.

The two MSBs of word three in each page shall contain data ID. Data ID number two (denoted by binary code 01) denotes the LNAV data structure of D(t) which is described in this Appendix and is the only valid value.

The Issue

First Recommended Clarification

20.1 Scope.



This appendix describes the specific legacy navigation (LNAV) data structure denoted by data ID number 2 for the lower set of PRN numbers (PRN 1-32). This data ID number, when transmitted as part of the LNAV data, shall be represented by the two-bit binary notation as 01. Data ID number 1 is no longer in use. The LNAV data structure for the lower set of PRN numbers is denoted as LNAV-L. The LNAV data structure for the upper set of PRN numbers (LNAV-U) is described in Appendix IV.

20.1 Scope.

Should Be

This appendix describes the specific legacy navigation (LNAV) data structure denoted by data ID number 2 for the lower set of PRN numbers (PRN 1-32). This data ID number, when transmitted as part of the LNAV data, shall be represented by the two-bit binary notation as 01. Data ID numbers 0, 1, and 3 are not defined in this IS is no longer in use. The LNAV data structure for the lower set of PRN numbers is denoted as LNAV-L. The LNAV data structure for the upper set of PRN numbers (LNAV-U) is described in Appendix IV.

SPACE STEER COMMAND

APPROVED FOR PUBLIC RELEASE Same Clarification, Second Occurrence

40.1 Scope



This appendix describes the specific legacy navigation (LNAV) data structure denoted by data ID number 2 for the upper set of PRN numbers (PRN 33-63). This data ID number, when transmitted as part of the LNAV data, shall be represented by the two-bit binary notation as 01. Data ID number 1 is no longer in use. The LNAV data structure for the upper set of PRN numbers is denoted as LNAV-U. The LNAV data structure for the lower set of PRN numbers (LNAV-L) is described in Appendix II.

40.1 Scope.

Should Be

This appendix describes the specific legacy navigation (LNAV) data structure denoted by data ID number 2 for the upper set of PRN numbers (PRN 33-63). This data ID number, when transmitted as part of the LNAV data, shall be represented by the two-bit binary notation as 01. Data ID numbers 0, 1, and 3 are not defined in this IS no longer in use. The LNAV data structure for the upper set of PRN numbers is denoted as LNAV-U. The LNAV data structure for the lower set of PRN numbers (LNAV-L) is described in Appendix II.

SPACE

Second Recommended Clarification

20.3.3.5.1.1 Data ID and SV ID.

Currently Is

The two MSBs of word three in each page shall contain data ID. Data ID number two (denoted by binary code 01) denotes the LNAV data structure of D(t) which is described in this Appendix and is the only valid value.

20.3.3.5.1.1 Data ID and SV ID.

Should Be

The two MSBs of word three in each page shall contain data ID. Data ID number two (denoted by binary code 01) denotes the LNAV data structure of D(t) which is described in this Appendix and is the only valid value <u>for this Appendix</u>.

Same Clarification, Second Occurrence

40.3.3.5.1.1 Data ID and SV ID.

Currently Is

The two MSBs of word three in each page shall contain data ID. Data ID number two (denoted by binary code 01) denotes the LNAV data structure of D(t) which is described in this Appendix and is the only valid value.

40.3.3.5.1.1 Data ID and SV ID.

Should Be

The two MSBs of word three in each page shall contain data ID. Data ID number two (denoted by binary code 01) denotes the LNAV data structure of D(t) which is described in this Appendix and is the only valid value <u>for this Appendix</u>.

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QUESTIONS & COMMENTS?



Walk-on Topics



Action Item Review

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Closing Remarks



Backup Slides



Acronyms

		1	
AFL	Available for Launch	IBR	Integrated Baseline Review
ASIC	Application Specific Integrated Circuit	IDR	Implementation Design Review
CDD	Capability Development Document	JTLV	Joint Light Tactical Vehicle
CDR	Critical Design Review	LCS	Launch and Checkout System
DAGR	Defense Advanced GPS Receiver	MGUE	Military GPS User Equipment
DDG	Arleigh Burke Guide Missile Destroyer	MSI	Miniature Serial Interface
DT	Developmental Testing	OCX	Operational Control System
FOT&E	Follow-on Operational Test and Evaluation	OT	Operational Testing
FQT	Formal Qualification Testing	PDR	Preliminary Design Review
FUE	Field User Evaluation	PNT	Positioning, Navigation, and Timing
GNST+	GPS IIIF Non-flight Satellite Test Bed	SIS	Signal in Space
GRAM–S/M	GPS Receiver Application Module –	TRV	Technical Requirements Verification
	Standard Elec Module/Modernized	URE	User Range Error
HH	Handheld	USAF	United States Air Force
HPE	Hewlett Packard Enterprise	USMC	United States Marine Corps
IBM	International Business Machines	USN	United States Navy